

CALIFORNIA STATE SCIENCE FAIR 2015 PROJECT SUMMARY

Name(s)	Project Number
Benjamin Kolland	
	35357
Project Title	55557
Separation Anxiety: A Plan for Safer Model Rocket Deployment	
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Abstract (Cools	
Objectives/Goals Current model rocket deployment systems are limited to single motor ejection	arget or separate
explosive charges controlled by electronics. The purpose of this projectis to de	ign a non-pyro parachute
deployment system (because minors cannot handle regulated explosives).	
Methods/Materials	\checkmark
A 90cm booster was constructed from 98mm cardboard tubing. The 110cm lon hatches actuated by model aircraft retract servos. An Eggimer attimuter I solde specified points in the rocket's flight. The final design's 40cm payload had an all servo released parachute compartment. The hatches were ground tested but whe	payload section had two
hatches actuated by model aircraft retract servos. An Eggimer altimater I solde	red triggered the servos at
specified points in the rocket's flight. The final design's 40cm payload had an al	timeter bay and single
servo released parachute compartment. The natches were ground tested but whe	parachuta after a small
multiple failures. The system was redesigned to use gravity to release the main drogue parachute was ejected at apogee using an explosive charge built into the	motor
The initial design testing went well, so the hatch setur was flown. However, the	e hatches didn't stay closed
during boost or deploy as planned. A new design was built and first RC ground tested, and then	
flight-tested to 166m with the main parachute RC released a approximately 75m. The motor eject worked	
The initial design testing went well, so the hatch setup was flown. However, the hatches didn't stay closed during boost or deploy as planned. A new design was built and first RC ground tested, and then flight-tested to 166m with the main parachute RC released as approximately 75m. The motor eject worked to deploy the drogue parachute, and the gravity release design for the main parachute worked perfectly. Three flights using Eggtimer altimeter control were not successful due to altimeter problems, but debugging and five more vacuum torm confirmed the sum of the gravity more was functional and deployed the main	
Three flights using Eggtimer altimeter control were not successful due to altimeter problems, but debugging and five more veguum too confirmed the given was functional and deployed the main	
debugging and five more vacuum tests confirmed the system was functional and deployed the main release at 45m. Two successful flights to 148m and 15 m using Eggtimer altimeter control deployed the main parachute as planned at 45m. A high altitude flight to 782m with winds of 12kph landed within 100m of the launch site after successful main deploy at the planned altitude of 102m. Two final flights deployed the planned arely due to foilure from equilar fortigue.	
main parachute as planned at 45m A high althude flight to 782m with winds of 12kph landed within	
100m of the launch site after successful main deploy at the planned altitude of 102m. Two final flights	
deployed early due to failure from coupler fatigue.	
Conclusions/Discussion	
My initial design failed due to gravity and aerodynamic issues, so I debugged and built a new design. It worked after some additional modification to altimeter programming. I ran multiple static (vacuum) and	
worked after some additional modifications to altimeter programming. I ran multiple static (vacuum) and flight tasts, which showed variance is a suite in the and I was able to build a non-pure deployment.	
flight tests, which showed very promising results. In the end, I was able to build a non-pyro deployment release for the main parachete that worked reliably and was easy to prep for flight. This system enables a	
lower cost, simple alternative to handling black powder charges for parachute d	eployment.
Summary Statement	
This project demonstrates an innovative design for model rocket dual deployme	ent using a safe,
non-pyrotechnic system.	
Help Received	
David Raimondi for suggestions based on his experience with parachute system	as Dave Cornelius for
failure analysis help on my initial design, my dad for driving to launches and ground support at launches.	